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# **LIMITATIONS**

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#### **SECTION 2**

#### **LIMITATIONS**

### 2.1 GENERAL

This section provides the "FAA Approved" operating limitations, instrument markings, color coding and basic placards necessary for the safe operation of the airplane and its systems.

Limitations associated with those optional systems and equipment which require handbook supplements can be found in Section 9 (Supplements).

### 2.3 AIRSPEED LIMITATIONS

SPEED	KIAS	KCAS
Never Exceed Speed $(V_{NE})$ - Do not exceed this speed in any operation.	183	186
Maximum Structural Cruising Speed $(V_{NO})$ - Do not exceed this speed except in smooth air and then only with caution.	146	148
Design Maneuvering Speed (V <sub>A</sub> ) - Do not make full or abrupt control movements above this speed.  At 2900 LBS. G.W.  At 1865 LBS. G.W.	119 96	121 96
CAUTION		
Maneuvering speed decreases at lighter weight as the aerodynamic forces become more pronounce interpolation may be used for intermediate grown Maneuvering speed should not be exceeded while crough air.	ed. Linear oss weights.	
Maximum Flaps Extended Speed ( $V_{FE}$ ) - Do not exceed this speed with the flaps extended.	103	104
Maximum Landing Gear Extension Speed - Do not exceed this speed when extending the landing gear.	129	130
Maximum Landing Gear Retraction Speed - Do not exceed this speed when retracting the landing gear.	107	109
Maximum Landing Gear Extended Speed (V <sub>LE</sub> ) - Do not exceed this speed with the landing gear extended.	129	130

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### 2.5 AIRSPEED INDICATOR MARKINGS

Ø.	MARKING	# 8	IAS
	Radial Line (Never Exceed) ow Arc (Caution Range - Smooth Air Only)	8 9	183 KTS 146 KTS to 183 KTS
Gree	en Arc (Normal Operating Range)		63 KTS to 146 KTS
Whi	te Arc (Flap Down)		56 KTS to 103 KTS
			* ×
POW	ER PLANT LIMITATIONS		
(a)	Number of Engines		1
(b)	Engine Manufacturer		Teledyne Continental
(c)	Engine Model No.		TSIO-360-F
(d)	Engine Operating Limits		a '
` ′	(1) Maximum Horsepower		200
	(2) Maximum Rotation Speed (RPM)		2575
	(3) Maximum Manifold Pressure (In. Mercury)		41
	(4) Maximum Oil Temperature		240°F
(e)	Oil Pressure		
	Minimum (red line)		10 PSI
	Maximum (red line)		100 PSI
(f)	Fuel Pressure		
	Maximum (red line)		19 <b>PS</b> I
(g)	Fuel Grade (minimum octane)		100LL - Blue or
(0)			100/130 - Green
(h)	Number of Propellers	8	1
(i)	Propeller Manufacturer		Hartzell
(j)	Propeller Hub and Blade Model		BHC-C2YF-1BF/F8459A-8R
(k)	Propeller Diameter		
()	Minimum	•	76 IN.
	Maximum		75 IN.
(1)	Blade Angle Limits		20
(*/	Low Pitch Stop		14.2 + .2°
	III.i. Dital. Cham		20 + 10

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29 ± 1°

High Pitch Stop

### 2.9 POWER PLANT INSTRUMENT MARKINGS

(a)	Tachometer		
	Green Arc (Normal Operating Range)		500 to 2000 RPM and
	<b>T</b>		2200 to 2575 RPM
	Yellow Arc (Avoid continuous operation above		
	32" Hg. manifold pressure)		2000 to 2200 RPM
(b)	Red Line (Maximum Continuous Power) Oil Temperature		2575 RPM
(0)	Green Arc (Normal Operating Range)		100° 4- 040° E
	Red Line (Maximum)		100° to 240° F 240° F
(c)	Oil Pressure		240 F
	Green Arc (Normal Operating Range)		30 PSI to 80 PSI
	Yellow Arc (Caution Range) (Idle)	3	10 PSI to 30 PSI
	Yellow Arc (Caution Range) (Start and Warm Up)		80 PSI to 100 PSI
	Red Line (Minimum) Red Line (Maximum)		10 PSI
(d)		· · · · · · · · · · · · · · · · · · ·	100 PSI
(-)	Green Arc (Normal Operating Range)		3.5 PSI to 19 PSI
	Red Line (Maximum)		19 PSI
(e)	Exhaust Gas Temperature (EGT)		12131
	Green Arc (Normal Operating Range)	4 4	1200° F to 1650° F
(f)	Red Line (Maximum) Manifold Pressure		1650°F
(1)	Green Arc (Normal Operating Range)	•	10 737
	Red Line (Maximum)		10 IN. to 41 IN. HG.
	,		41 IN. HG.

### 2.11 WEIGHT LIMITS

(a) Maximum Weight 2900 LBS
(b) Maximum Baggage 200 LBS

### NOTE

Refer to Section 5 (Performance) for maximum weight as limited by performance.

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### 2.13 CENTER OF GRAVITY LIMITS

Weight Pounds	Forward Limit Inches Aft of Datum	Rearward Limit Inches Aft of Datum
2900	86.0	90.0
2240	78.0	90.0

#### **NOTES**

Straight line variation between points given.

The datum used is 78.4 inches ahead of the wing leading edge at the intersection of the straight and tapered section.

It is the responsibility of the airplane owner and the pilot to insure that the airplane is properly loaded. See Section 6 (Weight and Balance) for proper loading instructions.

### 2.15 MANEUVER LIMITS

No acrobatic maneuvers including spins approved.

### 2.17 FLIGHT LOAD FACTORS

(a) Positive Load Factor (Maximum)

3.8 G

(b) Negative Load Factor (Maximum)

No inverted maneuvers approved

### 2.19 TYPES OF OPERATIONS

The airplane is approved for the following operations when equipped in accordance with FAR 91 or FAR 135.

- (a) Day V.F.R.
- (b) Night V.F.R.
- (c) Day I.F.R.
- (d) Night I.F.R.
- (e) Non Icing

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#### 2.21 FUEL LIMITATIONS

(a) Total Capacity
 (b) Unusable Fuel
 The unusable fuel for this airplane has been determined

77 U.S. GAL
5 U.S. GAL

as 2.5 gallons in each wing tank in critical flight attitudes.

72 U.S. GAL

The usable fuel in this airplane has been determined as 36.0 gallons in each wing tank.

(d) Fuel remaining when the quantity indicators read zero cannot be used safely in flight.

### 2.23 OPERATING ALTITUDE LIMITATIONS

Flight above 20,000 feet is not approved. Flight up to and including 20,000 feet is approved if equipped with oxygen in accordance with F.A.R. 23.1441 and avionics in accordance with F.A.R. 91 or F.A.R. 135.

#### 2.25 NOISE LEVEL

The noise level of this aircraft is 68.8 dBA.

No determination has been made by the Federal Aviation Administration that the noise levels of this airplane are or should be acceptable or unacceptable for operation at, into, or out of, any airport.

The above statement not withstanding the noise level stated above has been verified by and approved by the Federal Aviation Administration in noise level test flights conducted in accordance with FAR 36, Noise Standards - Aircraft Type and Airworthiness Certification. This aircraft model is in compliance with all FAR 36 noise standards applicable to this type.

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#### 2.27 PLACARDS

In full view of the pilot:

"THIS AIRPLANE MUST BE OPERATED AS A NORMAL CATEGORY AIRPLANE IN COMPLIANCE WITH THE OPERATING LIMITATIONS STATED IN THE FORM OF PLACARDS, MARKINGS AND MANUALS."

THIS AIRCRAFT APPROVED FOR NIGHT IFR NON-ICING FLIGHT WHEN EQUIPPED IN ACCORDANCE WITH FAR 91 or FAR 135.

In full view of the pilot, the following takeoff and landing check lists will be installed:

### TAKEOFF CHECK LIST

Fuel on Proper Tank Auxiliary Fuel Pump - Off
Engine Gauges - Checked
Alternate Air - Closed
Seat Backs Erect

Mixture - Set	
Propeller - Set	
Fasten Belts/Harness	

Flaps - Set	
Trim Tab - Set	
Controls - Free	
Doors - Latched	
Air Conditioner - 0	Off

### LANDING CHECK LIST

Fuel on Proper Tank
Seat Backs Erect
Fasten Belts/Harness

Auxiliary Fuel Pump - Off Mixture - Rich Propeller - Set Gear Down (129 KIAS Max) Flaps - Set (103 KIAS Max) Air Conditioner - Off

The "AIR CONDITIONER OFF" item in the above takeoff and landing check lists is mandatory for air conditioned aircraft only.

On the instrument panel in full view of the pilot:

MANEUVERING SPEED 119 KIAS AT 2900 LBS. (SEE P.O.H.)

On the instrument panel in full view of the pilot:

"DEMONSTRATED CROSSWIND COMPONENT 17 KTS"

On the instrument panel in full view of the pilot:

"NO ACROBATIC MANEUVERS, INCLUDING SPINS, APPROVED."

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On instrument panel in full view of the pilot:

"GEAR DOWN
"GEAR UP

129 KIAS (MAX)"
107 KIAS (MAX)"

"EXTENDED

129 KIAS (MAX)"

Near emergency gear lever:

"EMERGENCY DOWN"

"OVERRIDE ENGAGED AUTO-EXT-OFF LOCK PIN ON SIDE TO ENGAGE OVERRIDE: PULL LEVER FULL UP, PUSH LOCK PIN TO RELEASE OVERRIDE: PULL LEVER FULL UP & RELEASE"

Near gear selector switch:

"GEAR UP
"DOWN

107 KIAS MAX" 129 KIAS MAX"

Adjacent to upper door latch:

"ENGAGE LATCH BEFORE FLIGHT"

On the instrument panel in full view of the pilot:

"WARNING – TURN OFF STROBE LIGHTS WHEN IN CLOSE PROXIMITY TO GROUND OR DURING FLIGHT THROUGH CLOUD, FOG OR HAZE."

In full view of the pilot, in the area of the air conditioner controls when the air conditioner is installed:

"WARNING — AIR CONDITIONER MUST BE OFF TO INSURE NORMAL TAKEOFF CLIMB PERFORMANCE."

On inside of baggage compartment door:

"BAGGAGE MAXIMUM 200 LBS. SEE WEIGHT AND BALANCE DATA FOR BAGGAGE LOADING BETWEEN 150 LBS AND 200 LBS."

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### Adjacent to fuel tank filler caps:

"FUEL - 100/130 AVIATION GRADE - MIN. USABLE CAPACITY 36 GAL."

"USABLE CAPACITY TO BOTTOM OF FILLER NECK INDICATOR 25 GAL."

## Above fuel quantity gauges:

"FUEL REMAINING WHEN QUANTITY INDICATOR READS ZERO CANNOT BE USED SAFELY IN FLIGHT."

On the instrument panel in full view of the pilot:

"AVOID CONTINUOUS GROUND OPERATION 1700-2100 RPM IN CROSS/TAIL WIND OVER 10 KTS."

"AVOID CONTINUOUS OPERATION 2000-2200 RPM ABOVE 32" MANIFOLD PRESSURE."

### On the aft baggage closeout:

"MAXIMUM BAGGAGE 200 LBS. NO HEAVY OBJECTS ON HAT SHELF."

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